

The NASA Global Hawk UAS and its application for atmospheric science

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Aura Science Validation Goals:

The priorities for Aura validation during GloPac are:

- Measurements in the upper troposphere-lower stratosphere (UTLS) of:
- Temperature
- Carbon monoxide (CO) [MLS]
- Water vapor (H₂O) [MLS] • Ozone (O₃) [MLS, TES, OMI]
- Nitrous oxide (N₂O) [MLS] Aerosol size distributions in the UTLS

Measurements of:

- Column O₃ [OMI, TES]
- Sulfur dioxide (SO₂) [OMI] Formaldehyde (HCHO) [OMI]
- Nitrogen dioxide (NO₂) [OMI]

 Aerosol optical depth and aerosol layer heights Cloud height [OMI]

About the Aircraft:

Dimensions: 116.2 ft Wingspar 44.4 ft 15.2 ft Weight:

Payload > 1,500 lbs 9,100 lbs **Empty weight** 14,500 lbs Take-off fuel Take-off gross weight 25,600 lbs

Propulsion: Engine **Thrust**

AE-3007H Turbofan Flat rated at 7500 lbs thrust at sea level

31+ hrs

15 hrs

172 m/s

by payloads)

Unmanned Aircraft Systems (UASs) provide a new and exciting avenue for atmospheric observations.

7000 km range), the Altair (20 hrs., 6500 km), the Aerosonde (30 hrs., 3000 km), and the Global Hawk

(GloPac 2009). This mission will be flown on the NASA Global Hawk in the Spring/Summer of 2009

the long range, extended duration of the Global Hawk. The goals of the mission are to: 1) provide

validation observations for NASA satellites, 2) sample the break up of the Arctic stratospheric polar

The Global Hawk is a hybrid platform, melding extensive satellite-like geographical coverage with the

in-situ capability of a manned aircraft. GloPac 2009 will be the first atmospheric science mission to exploit

NASA has a number of unmanned aircraft systems. Among these are the Ikhana (24 hrs.,

(30 hrs, 22,000 km). In this poster we describe the upcoming Global Hawk Pacific Mission

vortex, and 3) observe cross-Pacific transport of aerosols and pollutants such as ozone.

Performance (demonstrated): Maximum Altitude Maximum Range Maximum endurance On-station endurance @ 4,600 nmi True airspeed @55 kft+ altitude True airspeed @25 kft altitude

Electrical Power – available to Payloads: DC (Engine-driven Generator) AC (Hydraulic-powered Generator)

Turn radius @ 55 kft+

WB-57 **Global Hawk** ER-2 65,000 ft (20 km) 70,000 ft (21 km) 60,000+ ft (18 km) 4,600 km 20,000+ km 5,600 km

8 hrs

210 m/s

6 hrs

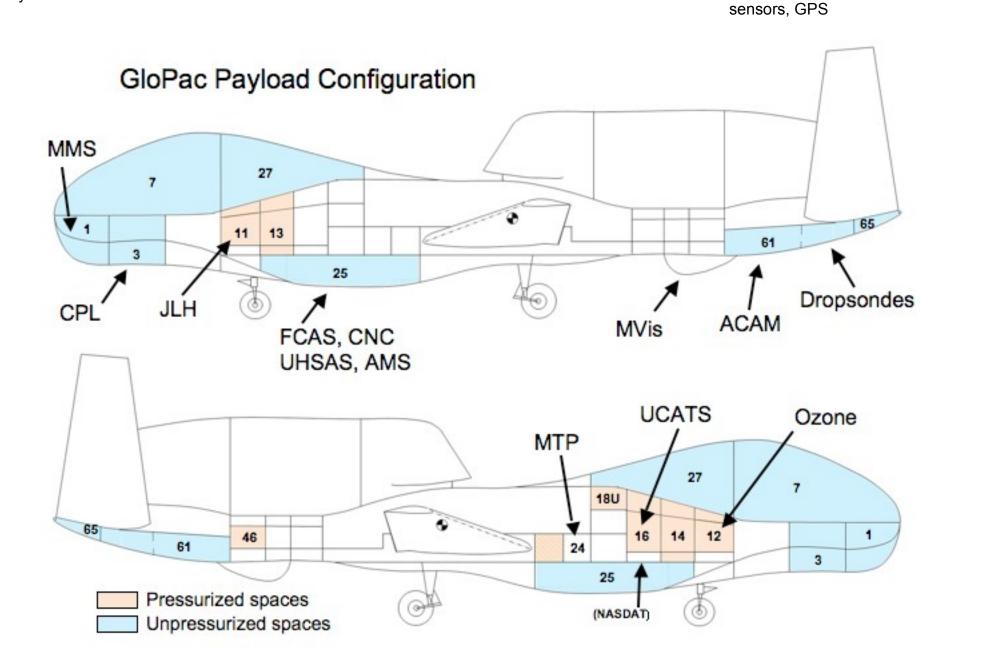
210 m/s

108 m/s 12 km 28 VDC, 72 A (2.0 KW) 115 VAC, 3-phase 400 Hz (8.2 KVA)

(AC converted to 28VDC as needed

About the Payload

Table 1. Global Hawk In situ Instruments	K <i>In Situ</i> Instru Technique	ument Payloa Payload location	d Principal Investigator	Institution r	Table 2. Global Hawk Remote instruments	Remote Inst	rument P Payload location	ayload Principal Investigator	Institution
Water vapor (H ₂ O):					Boundary layer nitrogen dioxi	de (NO ₂) retrieval:			
JPL Laser Hygrometer (JLH)	Tunable diode laser absorption	Fuselage (Zone 11)	R. Herman	Jet Propulsion Laboratory	Airborne Compact Atmospheric Mapper (ACAM)	Scanning spectrographs & cloud cameras	Rear fuselage (Zone 61)	S. Janz	NASA/ GSFC
Ozone (O ₃):				Laboratory		a cloud carricras	(Zone or)		
NOAA UAS Ozone	UV absorption	Fuselage (Zone 12)	R. Gao D. Fahey	NOAA/ ESRL	Cloud and aerosol properties:				
Long-lived gases:		(,	,		Cloud Physics Lidar (CPL)	3-wavelength	Forward	M. McGill	NASA/
Unmanned aircraft systems Chromatograph for Atmospheric	Gas chromatography	Fuselage (Zone 16)	J. Elkins	NOAA/ ESRL		backscatter lidar	fuselage (Zone 3)		GSFC
Trace Species (UCATS)					Vertical temperature profiles:				
Channel 1: N ₂ O, SF ₆ ;					Microwave Temp Profiler	Passive	Upper	M. Mahoney	.let
Channel 2: CO, H ₂ , CH ₄					(MTP)	microwave	fuselage	W. Wandicy	Propulsion
or CFC-11, CFC-12, Halon-1211					()	sensor	(Zone 22)		Laborator
Aerosol particles (0.008 - 2 μm	diameter):								
Condensation Nuclei	Supersaturation	Lower fuselage	J. Wilson	University	Atmospheric radiation (infrare		•		
Counter (CNC)	& growth	(Zone 25)		of Denver	Autonomous Modular Sensor	Scanning	Lower	J. Myers	NASA
	_				(AMS) System	multispectral	fuselage		Ames
Aerosol particles (0.09 - 1 μm (•					sensor system	(Zone 25)		
Focused Cavity Aerosol	Laser	Lower fuselage	J. Wilson	University	Flight path and scene content	documentation:			
Spectrometer (FCAS)	scattering	(Zone 25)		of Denver	MVIS Video Camera	Nadir-viewing	Lower	J. Myers	NASA
Aerosol particles (0.05 - 200 nm diameter):						color digital	fuselage	,	Ames
Ultra-High Sensitivity Aerosol	Laser	Lower fuselage	G. Kok	Droplet		video camera	(Zone 51)		
Spectrometer (UHSAS)	scattering	(Zone 25)	B. Gandrud	Measure-					
		(=====)		ment	Under development				
				Techno-	Pressure, temperature, relative	e humidity, and win	d vertical pro	ofiles to the s	urface:
				logy, Inc.	Dropsonde launch system	Release of	Rear	D. Fahey	NOAA/
Aircraft state parameters (pressure, temperature, winds):						disposable	fuselage		ESRL
Micrometeorological	Aircraft probes	Forward	P. Bui	NASA		sondes with	(Zone 61)	T. Hock	NCAR
Measurement	and inertial	fuselage		Ames		pressure,			
System (MMS)	navigation	(Zone 1)				temperature,			
	system					relative humidity			



In situ instrument scientific objectives

Water vapor (H2O) (JLH): Measure water vapor in the upper troposphere and lower stratosphere to (i) validate the Microwave Limb Sounder (MLS) on Aura, (ii) observe the distribution of ice supersaturation in and out of clouds, and (iii) identify and describe transport events in the lower stratosphere such as polar vortex remnants at midlatitudes.

Ozone (O₃) (NOAA UAS Ozone): Provide accurate O₃ measurements in the upper troposphere and lower stratosphere at 2-Hz sampling frequency for use in Aura validation and atmospheric composition and dynamics studies.

Long-lived gases (UCATS): Use measurements to (i) support Aura Satellite Validation of trace gases (N₂O, CFCs, CO, CH₄), (ii) determine age-of-air in the lower stratosphere (using SF₆), (iii) estimate the total organic and inorganic chlorine and bromine budgets in the lower stratosphere, (iv) identify transported air containing Asian air pollution (using SF₆, CO, O₃, CFCs, halons), (v) identify polar vortex fragments (using N₂O, CFCs, CH₄), and (v) identify air that has been transported across the tropopause (N₂O, CFCs, CH₄).

Aerosols (CNC, FCAS, UHSAS): Measure particle size distributions in the upper troposphere and lower stratosphere to understand the roles of transport, new particle formation, removal by clouds in determining the aerosol abundance, (ii) study the variation with location of aerosol properties and their controlling processes, (iii) characterize the particle population that provides nuclei for cloud formation in the UT, and (iv) observe particle size distributions in pyro-cumulus plumes in the UT/LS.

Aircraft state parameters (MMS): Provide sciencequality state variables (pressure, temperature, winds) to (i) compare with meteorological forecasts and analyses, (ii) constrain calculations of kinetic parameters, (iii) compute saturation conditions for water, nitric acid, sulfuric acid, (iv) identify gravity wave structures in the upper troposphere and lower stratosphere, and (v) provide state parameters for other in situ instruments.

Remote instrument scientific objectives

Boundary layer NO₂ retrieval (ACAM): Provide (i) spatial scaling of boundary layer NO₂ plumes to support validation of the Aura OMI and model studies of the next-generation geostationary satellite sensors, and (ii) NO₂ diurnal variability for model improvement for next-generation geostationary sensors.

Clouds and aerosol properties (CPL): Provide highaltitude profiling of the vertical structure of aerosols and clouds, (ii) identify cloud-free/elevated aerosolfree regions, (iii) identify type, height, coverage, and composition of clouds, (iv) validate Aura OMI aerosol optical depth and cloud height retrievals, and (v) validate Aura HIRDLS and MLS retrievals of cloud/ aerosol height and cloud phase.

Temperature profiles (MTP): Provide accurate vertical temperature profiles that can be used to (i) define the meteorological context along the flight track (e.g., aircraft location with respect to the tropopause) for other onboard measurements of trace gases and aerosols, (ii) validate temperature profile measurements made by Aura MLS and TES instruments, (iii) identify isentropes (i.e., surfaces of constant potential temperature) from the temperature curtain along the Global Hawk flight track so that atmospheric dynamics can be studied and transport pathways can be identified.

Atmospheric radiation (AMS): Provide (i) spatial maps of infrared emission and solar reflection, (ii) the spatial distribution of upper tropospheric and total water vapor (6.7 µm band), (iii) (under development) cloud micro-physical properties, using GSFC MODIS-MAS algorithms (optical thickness, effective radius, etc), (iv) horizontal structure of atmospheric aerosols, and (v) sub-visible cirrus.

Flight path documentation (MVIS): Provide flight path and scene content documentation along the

Pressure, temperature, and wind profiles (Dropsonde Launch System) (Under development): The dropsonde units will measure key meteorological and state variables in a vertical profile to the surface below the aircraft launch point (pressure, temperature, relative humidity, horizontal winds). These data will be used in a variety of ways: climate change detection, hurricane reconnaissance, atmospheric river forecasting, and satellite calibration and validation activities.

NASA Dryden Flight Research Center Photo Collection

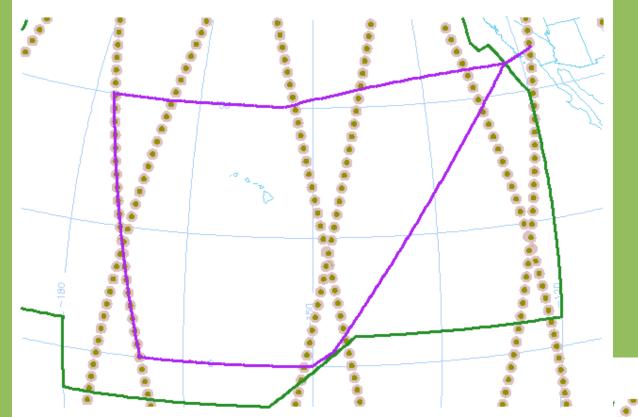
http://www.dfrc.nasa.gov/Gallery/Photo/index.html NASA Photo: ED07–0244–78 Date: December 3, 2007 Photo By: Tony Landis In its new white–and–blue NASA livery, an early development model of the Global Hawk unmanned

aircraft rests on the ramp at the Dryden Flight Research Center.

Sample GloPac Flights

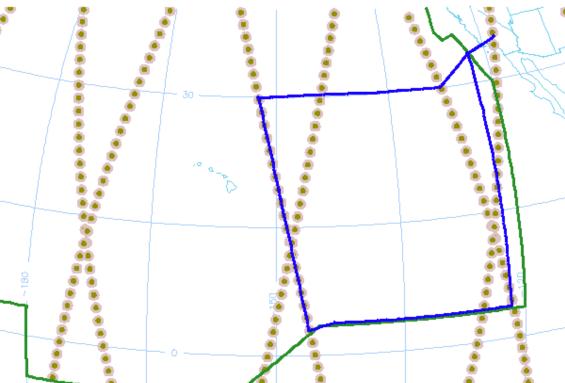
Six basic flight paths are shown below. The first five are designed to follow Aura or A-Train satellite sampling tracks, and the sixth is a flight into the polar region. The Aura Microwave Limb Sounder (MLS) sampling points are shown as gray-and-yellow dots; FIR boundaries (Oakland Oceanic) are shown in dark green. Interception flight tracks are shown for a maximum-duration flight. Each flight includes one to three descents to 45,000 ft (~ 14 km) for vertical profiling in situ.

Which flight gets flown on a given date will be determined by the overall dynamical situation as well as operational concerns. (The first flights will be 24 hours or less, with the full 30-hour flights coming later in the mission.)



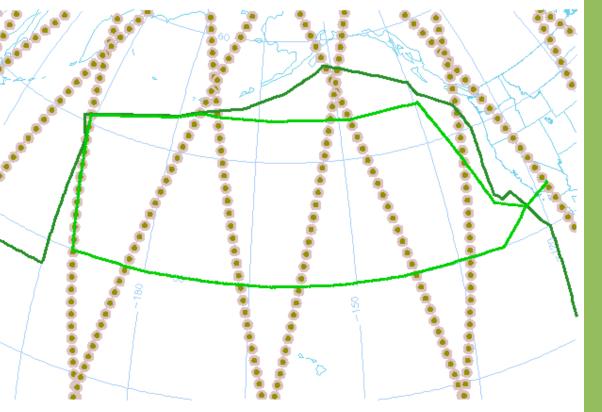
Flight 1 (purple) is typical of these flight plans. The aircraft heads out along one latitude (30 N) until it intercepts an MLS track (or other Aura or A-train track). It follows the track until it reaches the equator, which it then follows for a while before heading home. Total flight time is up to 30 hours. The along-track portion of the flight will be adjusted to intercept the track for the flight day.

Flight 2 (blue) is like Flight 1, but it is shorter (up to 24 hours duration). The aircraft will go out along latitude 30 N and return at about 4 N (the equator is outside the Oakland Oceanic area boundary here).



Flight 3 (brown) is back to a 30-hour flight, but it trades longitudinal range for latitudinal range. Unlike Flight 1, it no longer reaches into the mid-Pacific, but its latitude legs are now at 50 N and 4 N.

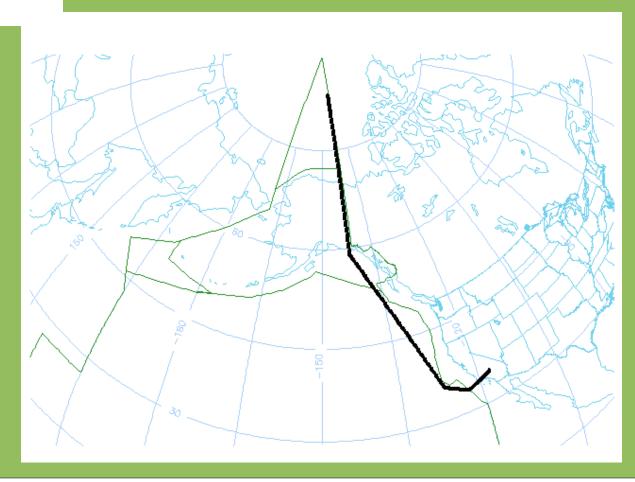
Flight 4 (bright green) covers a range of higher latitudes, from 50 N to 30 N. Consequently, a larger range of longitudes can be sampled in this 30-hour flight, all the way into the Western Pacific.



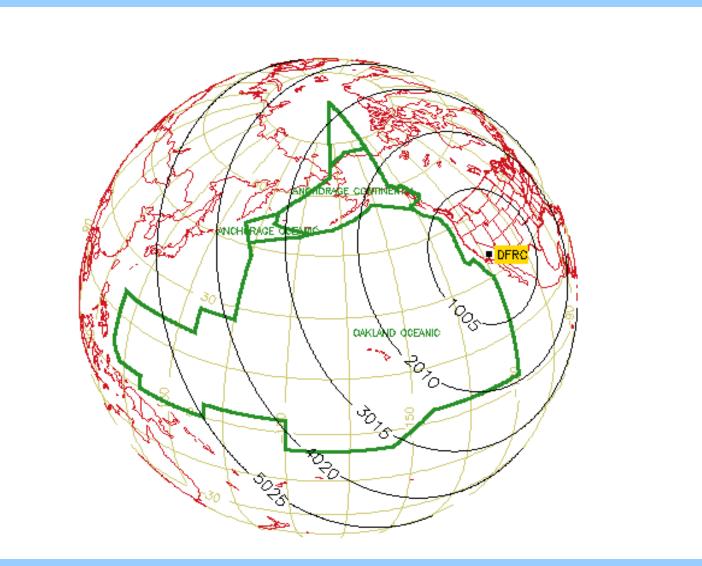
Flight 5 (red) is like Flight 4, except it is cut short to stay under 24 hours in duration.

Flight 6 (black) is a polar flight, cutting across Alaska to reach a latitude of 84 N before

returning to Dryden in California.



Area of Operations



Flights of unmanned aircraft face restrictions over populated areas, so most of the flights for GloPac will be over the ocean. To simplify interactions with air traffic control authorities, flights will generally take place within the aviation administrative area (Flight Information Area, or "FIR") known as "Oakland Oceanic", shown here in dark green. Also shown are range rings centered about NASA's Dryden Flight Research Center (DFRC), spaced 1005 nautical miles apart (approx. 1860 km), or about every three hours of flight time. The 5025 nmi circle corresponds to 15 hours one-way, or a 30-hour round trip.